

1 **Listing of the Claims**

2 **In the Claims:**

3 None of the claims have been amended in this response, but are presented here for the
4 convenience of the Examiner.

5 1. (Previously Presented) A method for enabling a user to create or modify a design for an
6 aircraft and evaluate flight characteristics of the design, comprising the steps of:

7 (a) enabling the user to input a plurality of parameters that define the design of the
8 aircraft;

9 (b) processing the plurality of parameters to generate a plurality of aerodynamic
10 coefficients that define a flight model for the design of the aircraft;

11 (c) producing a plurality of flight model data files that include:

12 (i) the aerodynamic coefficients generated; and

13 (ii) selected parameters input by the user; and

14 (d) enabling the user to evaluate the flight characteristics of the design by
15 interactive, simulated flying of the aircraft within a flight simulation program using the plurality of
16 flight model data files, wherein said interactive, simulated flying of the aircraft enables the user's
17 evaluation to be substantially based on a point of view of a pilot flying the aircraft.

18 2. (Original) The method of Claim 1, wherein the plurality of parameters include geometric
19 properties of the aircraft, and wherein the step of processing comprises the step of using the
20 geometric properties to determine a force developed by each of a plurality of component surfaces of
21 the aircraft.

22 3. (Original) The method of Claim 1, wherein the step of enabling the user to input the
23 plurality of parameters comprises the steps of:

24 (a) associating allowed limits for at least some of the plurality of parameters; and

25 (b) providing an indication if the user enters a parameter that is outside the
26 allowed limits associated with said parameter.

27 4. (Previously Presented) The method of Claim 3, further comprising the step of excluding
28 entry of any parameter that is outside the allowed limits associated with said parameter.

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1 5. (Original) The method of Claim 1, further comprising the steps of:

2 (a) enabling the user to modify at least one of a plurality of parameters of an
3 existing design for an aircraft; and

4 (b) repeating steps (b) through (d) in Claim 1 in regard to the design of the existing
5 aircraft as thus modified by the user, to enable the user to evaluate flight characteristics of the design
6 of the existing aircraft as thus modified by the user.

7 6. (Original) The method of Claim 1, further comprising the step of modifying the flight
8 model data files that were generated, based upon empirical data determined from use of the flight
9 simulation program, to increase an accuracy of the flight characteristics experienced when simulating
10 flying of the aircraft in the flight simulation program.

11 7. (Original) The method of Claim 1, wherein the step of producing the flight model data files
12 includes the step of producing one flight model data file that includes binary data defining the aerodynamic
13 coefficients for a plurality of component surfaces of the aircraft.

14 8. (Original) The method of Claim 1, wherein the step of producing the flight model data files
15 includes the step of producing one flight model data file that includes at least some of the
16 plurality of parameters input by the user that do not directly affect the flight characteristics of the
17 aircraft.

18 9. (Original) The method of Claim 8, wherein said one flight model data file defines at least
19 one of:

20 (a) an instrument panel configuration for the aircraft; and
21 (b) a plurality of sounds experienced when simulating flying of the aircraft.

22 10. (Original) The method of Claim 1, wherein the step of processing the plurality of
23 parameters comprises the steps of:

24 (a) determining an order in which the aerodynamic coefficients for specific
25 component surfaces of the aircraft are generated; and

26 (b) generating the aerodynamic coefficients for the component surfaces in the
27 order determined.

28 11. (Original) A memory medium on which are stored machine instructions for carrying out
29 the steps of Claim 1.

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1 12. (Previously Presented) A method for enabling a user to create or modify a design for an
2 aircraft and evaluate flight characteristics of the design as created or modified by the user, comprising
3 the steps of:

4 (a) enabling the user to input a plurality of parameters that define the design of the
5 aircraft;

6 (b) processing the plurality of parameters to generate a plurality of aerodynamic
7 coefficients in a predefined sequence, said predefined sequence being selected to ensure that any
8 aerodynamic coefficients required to generate other aerodynamic coefficients of component surfaces
9 are generated first, said aerodynamic coefficients being generated to define a flight model for the
10 design of the aircraft;

11 (c) producing flight model data in a format that is compatible with a selected flight
12 simulation program; and

13 (d) enabling the user to evaluate the flight characteristics of the design by
14 interactively simulating flying of the aircraft using the selected flight simulation program, said flight
15 simulation program using the flight model data, wherein said interactively, simulated flying of the
16 aircraft enables the user's evaluation to be substantially based on a point of view of a pilot flying the
17 aircraft.

18 13. (Original) The method of Claim 12, wherein the step of producing the flight model data
19 comprises the step of producing a binary flight model data file that includes the aerodynamic
20 coefficients for the aircraft and an environmental flight model data file that includes at least one of:

21 (a) a control panel configuration; and

22 (b) a set of sounds experienced by the user while simulating flying of the aircraft
23 with the flight simulation program.

24 14. (Previously Presented) The method of Claim 12, further comprising the steps of:

25 (a) enabling the user to modify a plurality of parameters for an existing aircraft
26 design; and

27 (b) carrying out steps (b) through (d) in Claim 12 for the existing aircraft design as
28 thus modified.

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1 15. (Original) The method of Claim 12, further comprising the steps of:

2 (a) associating allowed limits for at least some of the plurality of parameters; and
3 (b) excluding entry of any parameter that is outside the allowed limits associated
4 with said parameter.

5 16. (Original) The method of Claim 12, further comprising the step of modifying the flight
6 model data that was generated, based upon empirical data determined from use of the flight
7 simulation program, to increase an accuracy of the flight characteristics experienced when simulating
8 flying of the aircraft in the flight simulation program.

9 17. (Original) The method of Claim 12, wherein an aerodynamic coefficient for aircraft lift curve
10 slope is generated before an aerodynamic coefficient for static longitudinal stability.

11 18. (Original) A memory medium on which are stored machine instructions for carrying out
12 the steps of Claim 12.

13 19. (Previously Presented) A system for enabling a user to create or modify a design for an
14 aircraft and evaluate flight characteristics of the design, comprising:

15 (a) a memory in which are stored machine instructions that define a plurality of
16 functions;

17 (b) a display;

18 (c) a user input device for input of data, commands, and for controlling the
19 aircraft; and

20 (d) a processor that is coupled to the memory, the display, and the user input
21 device, said processor executing the machine instructions to carry out the plurality of functions,
22 including:

23 (i) enabling the user to input a plurality of parameters that define the
24 design of the aircraft through the input device;

25 (ii) processing the plurality of parameters to generate a plurality of aerodynamic
26 coefficients that define a flight model for the design of the aircraft;

27 (iii) producing a plurality of flight model data files that include the aerodynamic
28 coefficients generated and selected parameters input by the user; and

29 (iv) enabling a user to evaluate the flight characteristics of the design by
30 simulating flying of the aircraft interactively in response to graphic images on the display, responsive

1 to controls and commands provided by the user with the input device, wherein the flight
2 characteristics of the design simulated during flying are based upon the plurality of flight model data
3 files.

4 20. (Original) The system of Claim 19, wherein the plurality of parameters include
5 geometric properties of the aircraft, and wherein the processor uses the geometric properties to
6 determine a force developed by each of a plurality of component surfaces of the aircraft.

7 21. (Original) The system of Claim 19, wherein the machine instructions cause the processor
8 to:

9 (a) associate allowed limits for at least some of the plurality of parameters; and
10 (b) provide an indication if a user enters a parameter that is outside the allowed
11 limits associated with said parameter.

12 22. (Original) The system of Claim 21, wherein the machine instructions cause the processor
13 to refuse entry of any parameter that is outside the allowed limits associated with said parameter.

14 23. (Previously Presented) The system of Claim 19, wherein the machine instructions further
15 cause the processor to:

16 (a) enable a user to modify at least one of a plurality of parameters of an existing
17 design for an aircraft; and

18 (b) repeat steps (b) through (d) in Claim 19 in regard to the design of the existing
19 aircraft as thus modified by the user, to enable the user to evaluate flight characteristics of the design
20 of the existing aircraft as thus modified by the user.

21 24. (Original) The system of Claim 19, wherein the machine instructions further cause the
22 processor to modify the flight model data files that were generated, based upon empirical data
23 determined for flight simulation, to increase an accuracy of the flight characteristics experienced
24 when simulating flying of the aircraft.

25 25. (Original) The system of Claim 19, wherein the machine instructions cause the processor
26 to produce one flight model data file that includes binary data defining the aerodynamic coefficients
27 for a plurality of component surfaces of the aircraft.

28 26. (Original) The system of Claim 19, wherein the machine instructions cause the processor to
29 produce one flight model data file that includes at least some of the plurality of parameters input by a user
30 that do not directly affect the flight characteristics of the aircraft.

1 27. (Original) The system of Claim 26, wherein said one flight model data file defines at
2 least one of:

3 (a) an instrument panel configuration for the aircraft; and
4 (b) a plurality of sounds experienced when simulating flying of the aircraft.

5 28. (Original) The system of Claim 19, wherein the machine instructions cause the processor
6 to:

7 (a) determine an order in which the aerodynamic coefficients for specific
8 component surfaces of the aircraft are generated; and

9 (b) generate the aerodynamic coefficients for the component surfaces in the order
10 thus determined.

11 29. (Original) A system for enabling a user to create or modify a design for an aircraft and
12 evaluate flight characteristics of the design, comprising:

13 (a) a memory in which are stored machine instructions that define a plurality of
14 functions;

15 (b) a display;

16 (c) a user input device for input of data, commands, and for controlling the
17 aircraft; and

18 (d) a processor that is coupled to the memory, the display, and the user input
19 device, said processor executing the machine instructions to carry out the plurality of functions,
20 including:

21 (i) enabling the user to input a plurality of parameters with the user input
22 device, to define the design of the aircraft;

23 (ii) processing the plurality of parameters to generate a plurality of
24 aerodynamic coefficients in a predefined sequence, said predefined sequence being selected to ensure
25 that any aerodynamic coefficients required to generate other aerodynamic coefficients of component
26 surfaces are generated first, said aerodynamic coefficients being generated to define a flight model
27 for the design of the aircraft;

28 (iii) producing flight model data; and

29 (iv) enabling the user to evaluate the flight characteristics of the design by
30 simulating flying of the aircraft interactively in response to graphic images on the display and

1 controls and commands provided with the input device, using the flight model data.

2 30. (Original) The system of Claim 29, wherein the machine instructions cause the processor
3 to produce a binary flight model data file that includes the aerodynamic coefficients for the aircraft
4 and an environmental flight model data file that includes at least one of:

5 (a) a control panel configuration; and

6 (b) a set of sounds experienced by the user while simulating flying of the aircraft.

7 31. (Previously Presented) The system of Claim 29, wherein the machine instructions further
8 cause the processor to:

9 (a) enable a user to modify a plurality of parameters for an existing aircraft design;

10 and

11 (b) carry out steps (b) through (d) in Claim 29 for the existing aircraft design as
12 thus modified.

13 32. (Original) The system of Claim 29, wherein the machine instructions further cause the
14 processor to:

15 (a) associate allowed limits with at least some of the plurality of parameters; and

16 (b) exclude entry of any parameter that is outside the allowed limits associated
17 with said parameter.

18 33. (Original) The system of Claim 29, wherein the machine instructions further cause the
19 processor to modify the flight model data that was generated, based upon empirical data determined
20 from other simulated flying, to increase an accuracy of the flight characteristics experienced when
21 simulating flying of the aircraft.

22 34. (Original) The system of Claim 29, wherein an aerodynamic coefficient for aircraft lift
23 curve slope is generated before an aerodynamic coefficient for static longitudinal stability.